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SPECIAL FEATURES

NUMBER
17

AERONAUTICAL WORK AT M.I.T.
CONVERTIBLE MONOPLANE-BIPLANE WINGS
THE EUROPEAN FLIGHT OF THE NORGE

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VOL. XXX

APRIL 26, 1926

No. 12

Legislation

THE BINGHAM BILL passed the Senate in December and the Merritt or Parker Bill passed the House earlier in April. The Bingham Bill was drawn along broad lines and offered only moderate fixing. The House Bill, which was passed in our March 22 issue, is much more elaborate and, upon the way, has very drastic regulations covering places, fires, and building facilities. By the time that this issue reaches our readers the two bills will have gone into a committee of the Senate and the House and a compromise bill will probably have been passed by Congress.

It was inevitable that, soon or later, legislatures governing aircraft would pass Congress. Aviation has consistently tried to spread knowledge about air law and has emphasized the dangers of over strict legislation. The series of letters printed in *Aviation* about air regulations were carefully studied by the members of the committees which framed the bills. These letters, which came in two lots to publish were sent directly to Washington. Furthermore, the figures of mileage flown by commercial pilots during 1935 were brought out before the House during the discussion of the bill. Whatever sort of legislation Congress decides to pass, *Aviation* feels that it has done its part in giving information and in trying to stimulate their thinking.

The law, when passed, is likely to be deemed satisfactory and harmonious, however, by many plain and striped sweaters. It should also be remembered that the public, as represented by Congress, has a right to demand protection against irresponsible flies. Also it would not be forgotten that legislation is imperative if our transport lines are to be established if the regulations do in practice prove harmonious, a united protest from the flies could bring a modification fast, and also, both Congress and the Department of Commerce desire to foster civilian aviation.

International Air Navigation

THE SPECTACULAR growth of aircraft during the war period brought home to Europe the necessity which would arise for some sort of uniform regulations for international air navigation. A subcommittee of the Peace Commission was established shortly after the war to go into the matter and lay down regulations which could be adopted by all countries.

It is hard for an American to realize the importance of such regulations and their scope. Our country is extremely large and it is a race that we wish to cross its borders even on an extended flight. In Europe, on the other hand, one can hardly get well up in the air before crossing the boundaries of another country. The regulations necessary are partly military and partly civilian. There are considerable military

detains, which are supposedly secret and which firms of an other country are not supposed to fly over. There are regulations to cover the flying of military planes over the territory of a neighboring state.

At the time of the Armistice and Peace Conference, when the main outlines of the *Alt* Convention were drawn, the anti-Fry phase, perhaps predominated, but, even from the earliest moments, certain conditions were necessary.

[illegible]

Canada has been most sincere and helpful in allowing American places to fly over its territory. Canada has signed the Air Conventions. Soon they may wish to encourage American aviators and they have a right to do so. The States defined Air Navigation Conventions came up before the Senate in 1919 and it has been lying dormant ever since. In the mean time, the signatories to the Convention, through their permanent committee, are modifying and improving the regulations. This is a task in which the United States should have a part and it is to be hoped that the Senate will again take the matter up.

The Senate is slowly inching that muzzifying the International Air Conference might be getting as tangled up with the League of Nations and so it has carefully avoided bringing up any discussion on the overboard. The House has, however, had the good sense to realize that, if some time would say not be so far in the future, air travel between the United States and Europe and South America might be a matter of common concern and that it would be well to frame the civil aeronautics regulations so that they would coincide in as general way with the international regulations on the subject. At present, this country issues flying licenses going to the West Indies or Cuba in the same manner as it treats some foreign countries. The ocean laws are extremely solid and complete and apply to all vessels, but the land laws for the pilot flying from one country to another are not so new but have been made up as the need of the case arose on the journey. The Senate, it might be said, should be satisfied.

Aeronautical Work at M.I.T.

*Aeronautics Course at the Massachusetts Institute of Technology Covers all Fields
Extensive Research and Test Work also Carried on.*

A REGULAR UNDERGRADUATE course in mechanical engineering, leading to the degree of Bachelor of Science in that subject, has just been started at the Massachusetts Institute of Technology by vote of the Institute authorities. A research course is thus added to the fifteen or various branches of engineering and science which the Institute already offered, and a new opportunity is offered to students desiring to prepare themselves for technical work in the mechanical industry or for mechanical research.

Undergraduate Dissertation

Although the probable cause is an accelerated engineering of M. T. V. has been left unclear for some time (two years) and its production are holding important positions in the industry. The authors of this paper have been able to ascertain that it is appreciated that there are opportunities to reduce the risk to accelerated engineering as well. The criteria was primarily for gradual and not risky by deliberate engineering. The authors of this paper have been able to ascertain that it is appreciated that there are opportunities to reduce the risk to accelerated engineering as well. The criteria was primarily for gradual and not risky by deliberate engineering. The authors of this paper have been able to ascertain that it is appreciated that there are opportunities to reduce the risk to accelerated engineering as well. The criteria was primarily for gradual and not risky by deliberate engineering.

The work of the first two years in the concentrated students will be very much like that for industrial engineers. In the third year, one or two concentrated subjects will be

included, especially a course in theoretical aerodynamics and one in rigging and structural maintenance work, according to present plans. The senior year will be devoted almost entirely to aeronautical work on airplane design and production methods, wind tunnel research, and aerodynamics and propellers, and each student will carry through an original piece of work, either a research in the wind tunnel, a laboratory study of some aerobically material on which wind-tunnel data are at present available, or a research, on aerodynamic data of that nature.

The steady growth of interest in an automated engineering education at M. I. T. is evidenced by the presence of thirteen distinct seminars in the course in explosive design—a course offered, as already noted, primarily for graduate students. As Army and Navy schools are also directed for accelerated work, and as the course in explosive design is taken by all the usual candidates in the graduate year of their study, in the Institute, the total number enrolled during the present year was about thirty.

Lesson 104. Tour vs the industry

During the brief session period immediately following the industry convention, the audience was segregated by an official card at inspection of nonconfidential work of interest to the Eastern states. Through the prompt co-operation of the industry and of Governmental departments, it was possible to cover a great deal of ground in a few days. The Carlin, Keating, Wright, Atlantic Aircraft, Sikorsky and Huff-Daland planes being among those visited, together with the Naval Aircraft Division and the laboratory of the National Advisory Committee on Aeronautics. For information made the whole time, and others joined the party in the case of the Eastern states, to the end of the summer of 1918. The aircraft displayed was so good that it is probable that such a trip will be an annual feature in the future.



A ground view of the small MIT, used toward 200 from the far end. Note the photograph was taken of electric cables carrying current to the furnace under here from installed underground.

The equipment of the Maandoumia Institute of Technology for the carrying out of aerogeological research and test work consists, in the main, of two main stands. These stands are built of the aluminum sections Vokma type and are similar in construction, one, however, being considerably larger than the other. The smaller of the two, built some years ago, is 4 ft. vertical diameter at the top, taking a diameter of 6 ft. at the base. The air circuit is set up by a rough-bore two at 7 x 6 in. diameter, which draws in the air from the top of the stand.

The bell-mouth inside rapidly diminishes from a diameter of 9 ft. 6 in. to the parallel portion, is a curved taper, where the exit is of conical form, 22 ft. 4 in. long, terminating at the fin and is a diameter of 7 ft. 3½ in. A hosepipe is inserted and this is composed of metal tubes 2 in. 3 in. diameter and 32 in. long and placed immediately before the parallel portion of the tunnel. Since these tubes are cylindrical, they do not fit the taper of the entrance cone exactly. This loose mesh serves the purpose of strengthening any cord which might otherwise break in the air stream as a result of the rotation of the fin.

Two platforms are arranged around the tunnel, one over the top and the other at the side. The latter gives access to the interior of the tunnel by means of a window hinged along its upper edge.

Measurements of forces upon models supported in the air stream are made by means of a standard N.P.L. type balance. The maximum wind speed obtainable in this tunnel is 66 m.p.h. at 7.5 ft. from the fan. At this speed the electric motor develops 11 hp.

There is also in use in the tunnel a roof balance for the movement of rolling and piling materials on wheels in the wind station, the model being, in this case, suspended in the tunnel by a system of wires.

The Loose Wind Towed

The large tunnel was built more recently. As has been previously stated, this tunnel is similar in design to the sandstone. The diameter of the working opening, which is parallel for 15 ft, is 7 ft. 6 in. The entrance cone, similar to that of the sand tunnel, is 15 ft in diameter and 20 ft



Diagram illustrating method of force and moment measurements by the force M/T beam.

bearing, thereby minimizing vibration. The balance was of the all-steel suspension type and similar to that employed by Dr. Pmedel at Göttingen. Reference to the diagram will make the arrangement clear.

Field Measurements

The case of the *torques* in the wire— ϕ , measured on A balance and ψ , measured on B balance, is the total vertical force on the model, in this case on aerial. The balance also reads a moment about the leading edge of the model. The data at the model is measured as the vertical component of two forces, one, the true horizontal force acting downstream, and the other, the reaction to the force exerted by the wire. The latter is the force measured on the force of the balance. The tension in ϕ is, therefore, a measure of the horizontal force on the model and is measured on the balance C. A calibration, of course, is necessary. The purpose of the wire is to carry a weight outside the tunnel over a small distance in order to— to measure tension in the 40 lb.

Rolling moments can be measured by linking balance A and reading a difference in tension between the two wires a, a' . A suitable balance arm being fitted which permits the axes aa' of the balance to rotate in the rock-stands plane.

Additional Assessment

Other types of symposiums have been designed and conducted for use in the large variety of special situations of research work. In addition to the student symposium and those work sessions carried on in the symposium laboratory, a very considerable amount of research work and test work is done in a wide variety of subjects involving all branches of aeronautics and astronautics, whether applied to the airplane, ship, helicopter, or in other fields in which aerodynamic theory is applicable.

All aerodynamic instruction and all research and experimental test work conducted in the Department of Aeronautics and Astronautics is carried on in the Ames Research Center, Moffett Field, California. The Director of the Department is Professor Edmund P. Warner.

Convertible Monoplane-Biplane Wings

Interesting Wind Tunnel Experiments on the Possibilities of Compound Wings.

By RANDOLPH F. HALL

MANY ATTEMPTS have been made to develop experimental wings in order to improve the airplane's performance. Unaltered, however, has not been confined to a departure from the most tried and the most successful wing contour.

A line of wing development has been followed by the writer which has proved interesting and encouraging. In the development an efficient and relatively deep section monoplane wing of high speed and climb characteristics is converted into a biplane to obtain an increased lift for slow speed performance. The principle was described in an article entitled, "Possible Loss of Aerodynamic Development," which appeared in the February 15 issue of *Aviation*.

Inefficient Gap/Chord Ratio

While it is true that biplane wings of small gap/chord ratio are very inefficient at low angles of incidence, the condition improves considerably for high incidence angles and a large per cent increase over a single wing of similar dimensions is obtained. This is evident from Fig. 4, showing biplane lift (left ordinate) correction for varying gap/chord ratios of a typical wing of high incidence. The slope of low angles has been assumed but the curve appears reasonable. From this curve, for a gap/chord ratio .15, a fifty per cent increase in lift over a single surface might be expected.

Highly efficient deep element wings have been proposed but their maximum lift value has not been superior to this airfoil for comparative purposes. With the exception of their adaptability to incidence and internal construction, an improvement in panel action if it is possible to use the additional depth advantageously, is, the surface, where a convertible wing is capable of attaining greater lift with the corresponding maximum speed qualifications. An increase in



FIG. 1 EXPERIMENTAL WING NO. 1



FIG. 2 EXPERIMENTAL WING NO. 2



FIG. 3 BIPLANE WING NO. 3

resistance for this condition would, in case of the simple power at maximum speeds, be of little consequence. Additional resistance there, at both angles, the resistance of low gap/chord biplane is small.

In addition to the convertible feature, the interpretation of data necessitates the comparison between the wing may be found to further improve the performance. Also, in all probability should function in the manner of the Hinder

type, but, in loading as shown the section. The section procedure, that of displacing air from above by ventral action has been attempted and discussed. Fig. 1 shows the wing in use and in Fig. 3 the results are plotted. These data are of interest may be observed from tests of the wing at

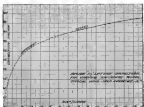


Fig. 4

Fig. 2 and the section curve of Fig. 5, although the maximum lift will still be at a lower angle. The permeability was very small and the maximum lift was low.

The only complete test of the convertible wing has been made upon the wing of Fig. 3. The lower surface of which is flattened. The curve of Fig. 6 relative to the section and Fig. 7, showing the characteristics curves for the wing as a biplane. These tests are not absolutely comparable because

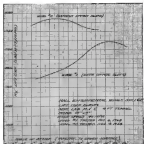


Fig. 5

of the lateral and drag values differences but are sufficient for relative comparison. The wings of the convertible biplane model were of the usual laminated wood construction and were covered by thin brass-plate running parallel to the chord

between the airfoils, the gap being about half the depth of the blade construction. The maximum lift of this model is 0.8025 at 15 deg incidence. The L/D value at high incidence was normal, showing the most gliding angle. The ratio of power at high incidence is at .575 at the chord area the leading edge, which is not far from the C.P. position of the lower surface. This advantage is most apparent in the structural design of the wing and in construction of airplanes.

Speed Range Possibilities

A comparison of the characteristics of this wing with several other well known airfoils is given in Table 1. While weight and simplicity are factors to merit comparison, the wing in question is not actually involved. The single leading edge and trailing edge tips require such be as seen, then the ordinary, which is not. Certainly large tail section of the rear tip, which, by means of interference, reduces the rear tip, can hardly be found extremely simple and favorable. In simpler aerodynamic this wing could be expected to permit a greater speed range than the one with the total wing, without service or other performance. The usual increase would depend upon the class of airplane. Ordinarily the increase would be of both incidence and velocity scale at the velocity scale, although, in comparison with very efficient high speed thin wing designs, the increase would be, at the best, and only where it is not needed and where it might mean the difference between being and being leading edge and specialized and correct pilots. From the example of Table 1 it follows that the leading speed and reduction

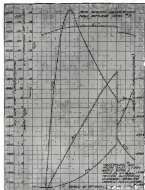


Fig. 6

is greater for high speed planes. The 15 m.p.h. difference between the convertible wing and the Hinder and is worth while.

The lateral control of a convertible biplane wing airplane may be accomplished in several ways. It may be incorporated in a part of the permeability control or separate reference may be provided. The latter may be located side-

probably of the convertible portion of the wing or carried as an air-braking trailing edge portion of the upper surface.

The proposed convertible wing has been outlined. The arrangement is intended to be easily modified and the wing tested, as indicated, to be improved or modified characteristics. A broad conversion that is a structural

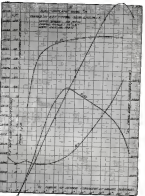


Fig. 7

arrangement is considered superior. It is not rather difficult at present what constitute limitations of practical design. The wing differs from the class of variable surface arrangements, in that the front elements are fixed and rigid. The only addition to an ordinary biplane is the narrow nose flap which is not complex. Dural bronze tail control is one of the standard forms of lateral control. The reference data supports between the two airfoils and leading and trailing edges, respectively, in weight. Supports would in the model tested. Profile deformation is as more likely to occur than in any wing with internal.

Comparison with detailed and adjustable flap wings should be made upon an applicable basis. Percentage increase in lift is a measure of an actual importance unless the maximum lift is considerably greater than that obtained by an ordinary wing. Otherwise the purpose of providing a lower leading speed is defeated. Obviously, to create from something inferior something normal, even though possibly a high percentage increase, is not necessarily an improvement. The ratio of maximum lift with the device to maximum drag obtainable is only a hypothetical consideration and not a comparable value for different wings. True comparison should be made between the ratio of desired lift values of wings in flight arrangement and the corresponding lift values for those Key values. The solution of a particularly advantageous characteristic from each of a number of independent wings cannot be fairly cited against any wing.

Subsequent model tests of the Bird No. 3 Convertible Monoplane-Biplane wing with a 2/3rd in slot at the upper surface, directly to the rear of the forward flap and parallel to the leading edge with a 1/16 in. slot in the type slot in the

been there—making against darkness in a Jeep with goggles, a head wind, and that I have prayed for night speed. But I have not. I would have reached my destination in a five-minute plane. I simply would have ditched the greatest speed and started later. At any rate, I certainly would not have suffered at all with, and against, rain, in my ideal conditions, for night speed.

Speed is only relative, anyhow. In the days when I got out last year in a single-engine Packard, by the way, leaving the highway at 35 m.p.h. was a exciting procedure. In the present day, or tomorrow, the Jeep, in anyone except the pilot of the machine, is dead. When 200 m.p.h. airplanes are common, the speed meters will read places that will go 300 m.p.h., in three or four.

A Matter of Speed

Early, while we were in still air, with the engine throttled down and just purred steadily in about an hour. The month means 300 m.p.h. at full throttle. The Jeep, I will acknowledge, with its riding speed of 70 m.p.h. and its true speed of about 60 m.p.h., is a lot slower. It is inevitable, however a head wind and flying low over the highway, to have the car owners below you hold you as speed. Thirty miles an hour more speed will be just enough to water them like a hot cat, besides giving you a little extra to back a head wind.

As to the type of plane, that is, a biplane or monoplane, I am partial to the monoplane. That is probably because I learned to fly in one, a biplane, and that is one of my early flights on a monoplane. Anyone who has ever flown a monoplane, however it is compared to a biplane. And looking at the matter purely from the esthetic standpoint, the monoplane has it over the biplane. In an esthetic sense, in 1915, I believe I moved far more airplane and prize money simply because I flew a beautiful French monoplane while my competitors had the most primitive biplane.

Personally, I like the appearance of the modern, three-wing monoplane. It is infinitely cheap to construct, most easily repaired than the biplane, is undoubtedly more efficient than the biplane, which, much better views, and readily adapts itself to all kinds of construction.

All aerial construction is the coming thing, without a doubt. Even those who are making flying-boat machines admit it. But, with them, I am not strong in the all-around plane of aerial construction. In other words, I do not see the wing type of aerial. I do not realize that to know enough about the light forces as present to just too much dependence on them at vital points. In my ideal monoplane, I would like to see a combination of aerial construction in a wooden fuselage not constructive to a whole class, and I would like the tail section and struts of metal, but not the wings. And, with the all-around construction in mind, I could just as well stick to the biplane. But I predict, here and now, that the all-around monoplane will be the "three" place of the future.

The Pilot's Comfort

Comfort for the pilot is something to which the airplane designer and builder has, up to the present time, given too little thought. I want here to say what I think is a good deal to which I can see far more in a time in comfort. A five-foot airplane designer often overlooks the fact that there are other people in the world who have given a little more time to him and who need more room than he has in an airplane cockpit. And, under this kind of comfort, I shall include visibility. I want to see where I am going without having to get my head out in the sky stream.

Concurrent design plans for anything from a sentence down to a couple and half, in a line of headwinds. I use my plane like most people use cars—to cruise around the country. And I want a place to just my comfort, instead of being allowed to sit in it, as with a biplane. With a three-wing design, some should be easily reached from the ground and should be provided with a good, strong look. A space for the tail should also be provided easily separated from the space in which a landing gear, or engine, are to be kept. I do not tell me to put my body in a box and keep them with a good clothes. Tail has got to stay.

There is a wild idea, speaking of comfort. Many a time, evening, I make my longings for the night about dark. I often dream the fact that a comfortable bed is a long way off, but that food may be obtained nearby. But what good is it to me to get a night's rest? So, I have often worked to a plane, from the place I landed, because of lack of transportation.

I have often wondered why an air machine could not be suspended under the fuselage of the plane so that, by letting down the back of the seat, a comfortable bed would be available. The engine from the passenger's cockpit would serve as a pillow. The pilot's head, of course, would be in the rear cockpit and his body protected by the fuselage. Four struts round the edge of the cockpit could be provided to control about steel tubes which would support a small canopy. Plenty of air, and complete protection from rain! Can you imagine "airplane comfort" in the future? I'd like to do it now. In fact, I often do, but I am much more interested in a Jeep.

The present tendency in the development of commercial aviation, is toward the big plane and the three-place or small twin-line machine. The snappy, two-place sportplane seems to have been neglected, at least in this country. In Europe, on the contrary, the sportplane has received considerable attention.

A Definite Example

The DeLavalized block in the outstanding sportplane in England. In fact, it's about the easiest little sportplane in the world, in my opinion. It is a two-place job with a four-cylinder, six-cylinder engine of 60 hp. And it has done a great deal to advance interest in aviation among a class of people who, up to its advent, took little interest in things aeronautical. Last airplane class here has formed all over England, and, in almost every case, the D.M. has been chosen as the official plane and an order placed for one or more.

Personally, I am fond of the opinion that the airplane will not really come into its own until the man in the street is known to be intimate with it. And I believe that the comfort man, and also the most satisfactory way of conducting the contact is by the introduction of what is popularly known as the "three" place. Undoubtedly the demand for such a plane exists and this demand will become greater and greater as more and more people take their friends for "three".

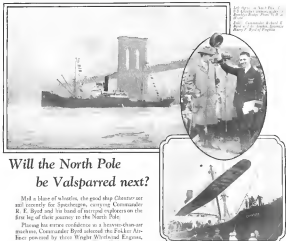
A New British Mooring Mast



The Conquest airplane mooring mast.

An end-up mooring mast, claimed to be the highest in the world, has been erected at Farnborough, near London, for the use of the two British airships, R.100 and R.101, now under construction for non-military purposes. The mast is 195 ft. high and is constructed of steel lattice work on a 60 ft. base. Since the top airship mooring platform was mounted on the windlights and in another the masts during raising and setting and the other serving as a balcony for ground crew—who are raised up the mast by an electric lift. A similar mast at Duxford is also serving comparable.

The masts, it will be recalled, are to be used in the situation of a passenger and mail, are constructed in a way between London and India, before it is extended to Australia.



Will the North Pole be Valsparred next?

Mid a fleet of whalers, the good ship *Chenier* set out recently for Spitzbergen, carrying Commander R. E. Byrd and his band of intrepid explorers on the first leg of their journey to the North Pole.

Placing his entire confidence in a heavy-duty air machine, Commander Byrd selected the Folker Airplane, powered by three Wright Whirlwind Engines to carry him over the Polar Regions. Under the extreme conditions which the ship is likely to see in the frozen North, the weathering and setting of the plane plays a most important part.

It is interesting to note in this connection that the shape, fuselage and tail surfaces of the Airplane are Valsparred—another indication of what Commander Byrd and the Folker Company think of Valspar!



Left: *Chenier* from Polar Expedition! The Folker Airplane is used in Cape Mendocino and California. Right: Byrd on his way across a polar flight in a Folker Airplane.

Foreign Development Association to seriously consider commercial aviation. That is a good sign, when such a world-renowned organization turns its attention to the airplane as a commercial vehicle.

Boys, with a Travel Air, dropped their last week and gave some of the local pilots a help in a real machine. The Travel Air is just getting into southern California and it is already making good. It is a fine little airplane. I have first hand, because Hoyt was good enough to let me take it up to 4,000 ft, slow and steady. It does most of the stunts easily, except that it hates to spin. Which is in its favor, considering from the economical standpoint.

Swallow Airplane Mfg. Co. News

Radio Division of Santa Barbara, Calif., who is still an enthusiastic flyer after 15 years experience, has signed a dollar contract with the Swallow Airplane Mfg. Co. of Wichita, Kan., for Southern California. The company delivered three standard Swallows to Mr. Douglas during the week of April 22.

Another new Swallow dealer is F. L. Bernhardt of Chicago. He has contacted for the territory included in the southern half of Illinois and the northern half of Wisconsin. Delivery of the first plane was made to Mr. Bernhardt on April 25.

Walter Z. Varney's new planes, which were found to be undersized for the Elton, St. Paul, Wash., route, are being re-equipped with Wright Whirlwind 500 hp. engines. The work of installation is being rushed at the Swallow factory.

Spokane, Wash.
Dr. E. Hovey Peterson

With a crowd of thousands watching and cheering, Phil Lane, Columbus, jumped off from plane at 6:22 a.m. on April 6, with the first bird of confidence we had to be started from Pocatello, Wash., to Elton, St. Paul, where it was to be loaded aboard the Government's transcontinental plane. Walter Z. Varney's Swallow with 500 engine took off with 407

lb. of mail and it arrived at its destination at 12:36 p.m.

Three national ground airplanes from Spokane, piloted by May John Foster and Capt. Tom Symons and Harold Neely, flew to Pocatello and unloaded the mail plane for a short distance on its mail run.

The starting of the national air mail route was made a ceremony for the entire West. An old-time Stageship left Spokane carrying mail on the methods of a half century ago. It was Spokane's "ritual" to launch the air mail and the route was given a warm welcome over its 500 mile journey. It required a week for the trip.

Newspaper photographers by the score took pictures of the ceremony and post office officials from Spokane, Seattle and other cities attended.

Chicago News

By One Glen

The Yankee Aircraft Co. reports that so far six of the Robertson Aircraft Company's D1V have come in on the new Air Mail route. The Yankee Company reports considerable activity at their field. Spring sales of ships have increased with leaps and bounds and only the other day the company sold one of their large Yankee Transporters to the Aerial Photograph Service, Inc. of the city. The Yankee Company has at present a number of ships in work and has excellent prospects for the coming season.

Le Perry-Cramer has taken the local agency for the West through the Midwest Airways Corp. of Memphis. The first ship was delivered last week at Adams Field where it will be permanently stored. Cramer has sold his Cuckoo to Thomas Schaefer, a student of the South Airplane Co.

Walter Mayne flew to St. Louis over the week end in his Liberty Super Standard. Meyer is still flying from Martins Grove through the mid and west men have evidently placed large orders upon all over the midwest field, probably to enable the flyer to find the right route to land in. Between the size posts and the midwest and severe, Walter Meyer is having a lively time logging his thousand out of the field.

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United States Air Forces

Air Service at Sequoy-Castaneda

The United States Army will participate in the Second Annual International Exposition at Philadelphia, Pa., by sending a detachment of troops with equipment representative of various arms and branches of the service. The Secretary of War has named Maj. Gen. Douglas MacArthur, Commanding General of the Third Corps Area, to be the War Department's representative at the Exposition. He has been directed to carry out the entire plan of the Department's participation. The troops sent to compose the Exposition Detachment are:

The Air Corps is to be represented by one composite squadron of 13 officers and 33 enlisted men which will be made up as follows: officers and 13 enlisted men from post stations from Bollinger Field, Mich.; 3 officers, 25 enlisted men and 3 bandmen from Langley Field, Va.; 2 officers and 5 enlisted men and 3 observation planes from Hadding Field, Washington, D.C.; 1 officer, 3 enlisted men and 3 observation planes from Mitchell Field, L. I. N. Y.; 1 officer and 1 under arms plane from McCook Field, Ohio; 2 officers, 13 enlisted men and 2 balloons from Aberdeen Proving Ground, Md.

Coordination in Buying Aircraft

An Aircraft Procurement Board to coordinate aerial purchases in the different departments of the government will be headed by the bill which Representative Venson (Dem.) of Texas, Ky., introduced April 14.

The board would be composed of the Chief of the Air Service of the Army, the Chief of the Bureau of Aeronautics of the Navy, and Assistant Secretaries of War, Navy, and Commerce. In addition to representing the heads of the different groups, it would be the duty of the board to outline a policy of procurement for the Federal Government.

Sets Limit of 2,500 Places

At a conference on aviation services with Secretary of War Duggan, the House Military Affairs Committee on April 14, decided upon the clarification of the present military aviation bill so as to limit definitely the number to 2,500 of airplanes on hand at the end of the present program.

The bill provides for increasing the air force to 2,500 planes. There was doubt as to the result of executive action as to whether this was inclusive or exclusive of 400 planes now being built under contract.

The Secretary of War was called in and informed the committee that the department really wants only 2,500 planes.

Enlisted Men to Fly

Eleven enlisted men of Campa Field, Bland, Ill., who were appointed flying cadets, left on March 20 to take primary flying training at Brooks Field, Texas, via: Staff Sgt. Leo F. Wood, Corporal Clarence W. Haddock and Louis R. McPherson, 1st Lt. Jack Adams, Pilot. Dale L. Brown, 1st Staff Sergeant, Everett V. Haddock, George M. Stinson, Dwight V. Haddock, O'Neil Young and Stanley C. Robbins.

Army Air Orders

Sgt. Lieut. Howard D. Davidson, A.S., Kelly Field, to South Field.

Sgt. Lieut. John L. Haddock, A.S., ordered from duty as A. S. Air. Fly. Sgt., Kelly Field, and will report to Com. O.E.

Sgt. Lieut. Ernest Stanley Mason, promoted to rank of First Lieut. Lieutenant Mason will remain in present duties.

Following officers of the A.S. are designated as students

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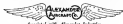
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